



Evaluation of Obturator and Sealing Cuff Properties for the M865 Training Projectile With Comparison to Ballistic Testing

by C. P. R. Hoppel, J. F. Newill,
and K. P. Soencksen

ARL-TR-2039

September 1999

19991008 064

Approved for public release; distribution is unlimited.

The findings in this report are not to be construed as an official Department of the Army position unless so designated by other authorized documents.

Citation of manufacturer's or trade names does not constitute an official endorsement or approval of the use thereof.

Destroy this report when it is no longer needed. Do not return it to the originator.

Army Research Laboratory

Aberdeen Proving Ground, MD 21005-5069

ARL-TR-2039

September 1999

Evaluation of Obturator and Sealing Cuff Properties for the M865 Training Projectile With Comparison to Ballistic Testing

C. P. R. Hoppel, J. F. Newill, and K. P. Soencksen
Weapons and Materials Research Directorate, ARL

Abstract

The nylon obturator and RTV sealing cuff for the M865 training round were evaluated to identify potential sources of ballistic variability associated with the material properties and material processing. While the properties of these materials are strongly dependent on processing conditions, temperature, and moisture content, the M865 performance variability is reduced by a well-engineered fracture mechanism that focuses the stresses in the obturator during sabot discard. A ballistic test was developed to validate the study. For the ballistic test, obturators were manufactured in "brittle," "tough," and "tough-wet" conditions. These three conditions produced significant differences in the mechanical properties (the maximum strength varied by a factor of 2, the elastic modulus varied by a factor of 25, and the elongation to failure varied by a factor of 10). However, the ballistic performance did not show any significant variability due to the obturator properties.

Acknowledgments

The authors would like to thank Mr. Ed Fennell, U.S. Army Armament Research Development and Engineering Center, and the M865 Process Variables Team for technical guidance and leadership throughout this project. The authors would also like to thank Dr. William Drysdale of the U.S. Army Research Laboratory (ARL) for his technical guidance on obturator performance. Also from ARL, Mr. Peter Dehmer, Dr. Travis A. Bogetti, Dr. Steven McKnight, and Dr. Nora Beck-Tan provided significant guidance on the properties of nylon materials; Dr. Peter Plostins led efforts to ensure that the ballistic tests were performed to meet the goals; and Mr. David Webb assisted in the design of the ballistic test as well as the data reduction from the test. Mr. Rollie Dohrn of Alliant Techsystems also contributed greatly to the success of this project by ensuring the proper manufacture and handling of the test projectiles. This work was funded through Mr. Robert Crawford, Rock Island Industrial Operations Command.

INTENTIONALLY LEFT BLANK.

Table of Contents

	<u>Page</u>
Acknowledgments.....	iii
List of Figures.....	vii
List of Tables	ix
1. Introduction	1
2. Nylon Obturator Band.....	2
2.1 Raw Material Properties.....	2
2.2 Processing.....	3
2.3 Environmental Effects.....	4
2.4 Stress Concentration Due to Notch and Geometry.....	8
3. Sealing Cuff.....	9
4. Ballistic Testing	11
4.1 Material Mechanical Properties.....	12
4.2 Environmental Conditions	12
4.2.1 Dry Out Testing	13
4.2.2 Moisture Absorption by the Obturators	14
4.2.3 Test Sample Preparation.....	14
4.3 Results	16
5. Conclusions	19
6. References	21
Bibliography	23
Distribution List	25
Report Documentation Page.....	37

INTENTIONALLY LEFT BLANK.

List of Figures

<u>Figure</u>	<u>Page</u>
1. Schematic Diagram of an M865 Projectile	11
2. Photograph of an M865 Projectile	2
3. Stiffness vs. Crystallinity for Nylon-610 Films (Kohan 1973)	3
4. Yield Point of Nylon-66 vs. Percent Crystallinity (Kohan 1973)	4
5. Tensile Stress-Strain Data for Nylon 6,6 at 23° C at 50% RH and Dry-as-Molded (DAM) Material Conditions (DuPont 1997)	5
6. Tensile Stress-Strain Data for Nylon 6,6 at 50% RH at Four Different Temperatures (DuPont 1997)	6
7. Flexural Modulus of Nylon 6,6 vs. Temperature at Various Moisture Contents (DuPont 1997)	6
8. Yield Stress Data for Nylon 6,6 Dry-as-Molded and 50% RH vs. Strain Rate and Temperature (DuPont 1997)	7
9. Effect of Temperature and Strain Rate on the Elastic Modulus of Nylon 6,6 at Two Moisture Levels (Kawahara, Brandon, and Korellis 1988)	7
10. Finite Element Model of Sabot Discard	8
11. Circumferential Stress in the Obturator During Discard for an Unnotched Obturator	9
12. Stress Distribution at the Forward Edge of the Sealing Cuff During Sabot Discard	10
13. Stress Distribution at the Bolts in the Sealing Cuff During Sabot Discard	11
14. Obturator Moisture Loss vs. Time	13
15. Percent Weight Gain vs. Time for Obturators Conditioned at 50% RH (Specimens 1988 C, 1988 D, 1997 C, 1997 D, and 1997 E) and 90% RH (Specimens 1988 A, 1988 B, 1997 A, and 1997 B)	14

<u>Figure</u>	<u>Page</u>
16. E3 Version of the M865 Projectile	16
17. Comparison of the Original M865 Projectile (Left) to the E3 Version (Right)	17
18. Obturator Pieces Found During the Ballistic Test	18
19. Close-up Photograph of Obturator Pieces Found During the Ballistic Test	18
20. Sabot From a Separate Ballistic Test Showing Soot in the Obturator Seat Due to Gas Leakage Underneath Obturator	19

List of Tables

<u>Table</u>	<u>Page</u>
1. Test Matrix (Number of Projectiles for Each Configuration)	12
2. Mechanical Properties of the Molded Test Projectiles	12
3. Average Obturator Mechanical Properties After Environmental Conditioning ...	15

INTENTIONALLY LEFT BLANK.

1. Introduction

This project was initiated to evaluate the materials used in the nylon obturator band and rubber sealing cuff for the M865 training projectile, to assess the ballistic implications of the material properties, and to evaluate the performance during ballistic testing. A literature search and a series of analyses were completed to evaluate the potential effects of variability in the raw material properties, processing effects, and environmental effects on the ballistic performance. The range of properties was then used in dynamic analyses (Newill et al., to be published) to predict the potential effects on the in-bore behavior on the projectile. Based on this study, an experimental program was designed to test the limiting values ballistically. The results of the study (in section 4) showed that the obturator material properties had little effect on ballistic performance.

A schematic of an M865 projectile is shown in Figure 1, and a photograph of the original version is shown in Figure 2.

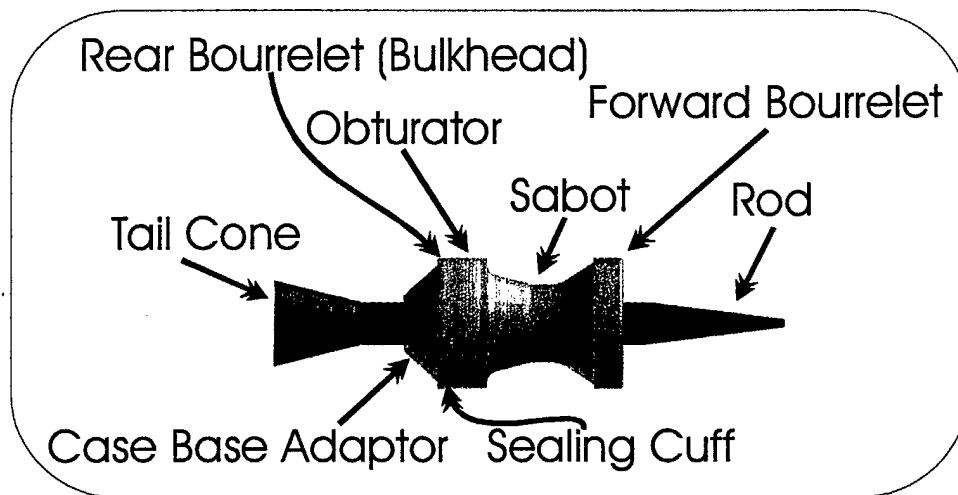


Figure 1. Schematic Diagram of an M865 Projectile.

The obturator band is located in the obturator seat on the rear bulkhead of the sabot. It is attached to the projectile with a knurled interface and helps hold the three sabot petals together. Notches are cut into the forward edge of the obturator and are aligned with the seams between sabot petals to initiate fracture of the obturator during discard. The M865 obturator is different from obturators on 120-mm tactical kinetic energy projectiles (M829, M829A1, and M829A2) in that it is broken during the discard process instead of at muzzle exit. The obturators on the M829, M829A1,

and M829A2 are broken by the loss of the support from the tube as the bullet exits the muzzle due to the large internal pressure. The sealing cuff on the M865 is located aft of the obturator and is designed to adhere to the sabot during discard, tearing along the seams between the petals after the obturator breaks. The sealing cuff also provides some sealing of the projectile during launch.

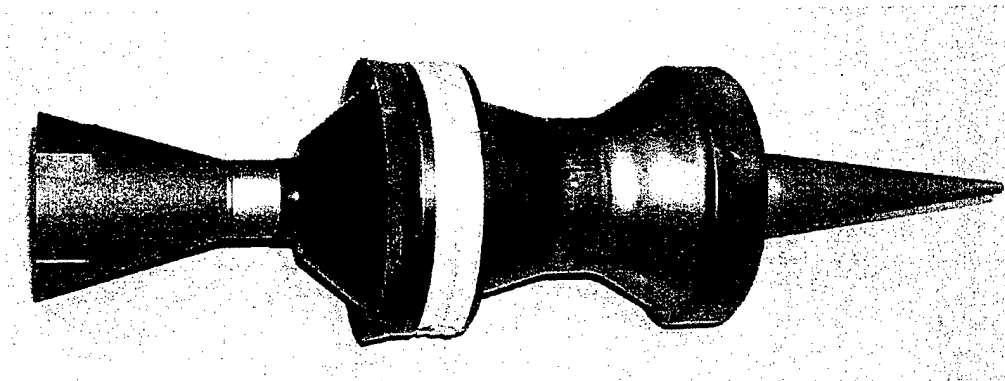


Figure 2. Photograph of an M865 Projectile.

Several problems involving the obturator and sealing cuff have occurred during production of the M865 projectile. The obturators have cracked during the final machining process, assembly, and handling of the projectile. Problems reported on the sealing cuff have involved occasional anomalies with discard. These problems have been attributed to poor interfacial adhesion between the sabot and sealing cuff.

2. Nylon Obturator Band

2.1 Raw Material Properties.

The nylon obturator band is made of injection-molded nylon 6,6. The specific nylon used for this program is DuPont Zytel 101. The raw materials are purchased to the specification for general-purpose nylon 6,6 in ASTM 4066-96a, "Standard Specification for Nylon Injection and Extrusion Materials (PA)" (ASTM 1996). The acceptance data from both of the contractors all met this specification and showed very low variability, indicating that raw material properties would have little influence on variability in the final molded obturator.

2.2 Processing.

The processing of the nylon obturators is much more significant in terms of variability in final properties. Two important aspects of processing are storage of the material prior to molding and material toughness. Material storage is important because it is critical that the nylon be protected from moisture prior to injection molding. Nylon is hydroscopic and will absorb moisture rapidly in ambient conditions. Any moisture in the nylon during the molding process will cause voids to be formed in the final part, making it brittle, or could damage the obturator's ability to seal.

Toughness in the material is also an important processing concern. It can be related to the amount of crystallinity and the structure of the crystals. In general, increasing the crystallinity makes the nylon more brittle. However, the structure of the crystals also is important. For equal amounts of crystallinity, small crystals produce a tougher microstructure than large crystals. The degree of toughness in the nylon can be controlled through the initial mold temperature and cooling cycle during the injection molding process. If the nylon part is cooled rapidly from the molding temperature, it will solidify into an amorphous structure before crystals form. If the material is cooled slowly, crystals will form in the nylon. The degree of crystallinity can then be adjusted by altering the cooling cycle.

The degree of crystallinity will affect the appearance and the mechanical properties of the nylon. An amorphous nylon can be translucent, or clear in color. In general, amorphous nylon will have a high degree of toughness, with a low elastic modulus and yield strength, and a high strain to failure. Increasing the crystallinity of nylon makes it more opaque (giving it a whiter color) and makes it more brittle. The brittleness increases the elastic modulus and yield strength and reduces the strain to failure of the material. Figure 3 shows how the stiffness of nylon varies vs. percent crystallinity for samples conditioned at three different moisture levels. The relationship between the yield strength of nylon 6,6 and percent crystallinity is shown in Figure 4. Both the elastic modulus and yield strength increases linearly with percent crystallinity.

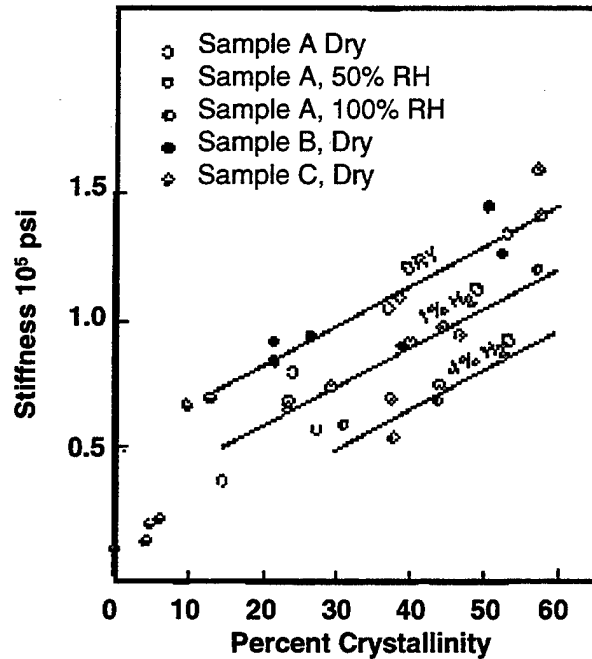


Figure 3. Stiffness vs. Crystallinity for Nylon-610 Films (Kohan 1973).

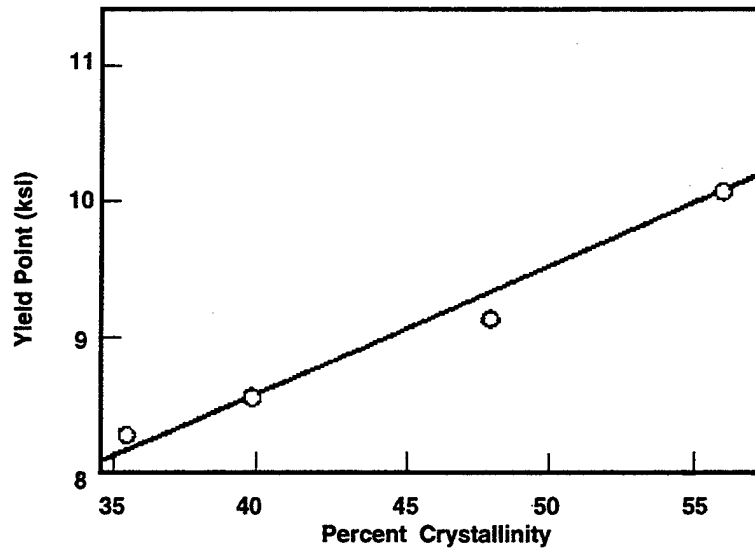


Figure 4. Yield Point of Nylon-66 vs. Percent Crystallinity (Kohan 1973).

2.3 Environmental Effects.

After nylon obturators are made, the moisture content and temperature can significantly influence their mechanical properties. As mentioned earlier, nylon is hygroscopic in nature and will absorb up to 8% moisture over time. Increasing the temperature of the specimens would greatly increase the rate of moisture absorption. In addition, increasing the RH levels would increase the amount of moisture that these specimens would gain since the saturation level of the material is

proportional to the exposed RH (Tsai 1988). The absorbed moisture will cause the nylon to swell through hygrothermal expansion.

Absorbed moisture will also change the mechanical properties of the nylon. Figure 5 shows stress-vs.-strain curves for nylon 6,6 in the dry-as-molded (DAM) condition and a specimen conditioned to 50% RH. Note that the dry specimen is much more brittle. It is stiffer and has a higher yield strength than the specimen preconditioned to 50% RH (DuPont 1997).

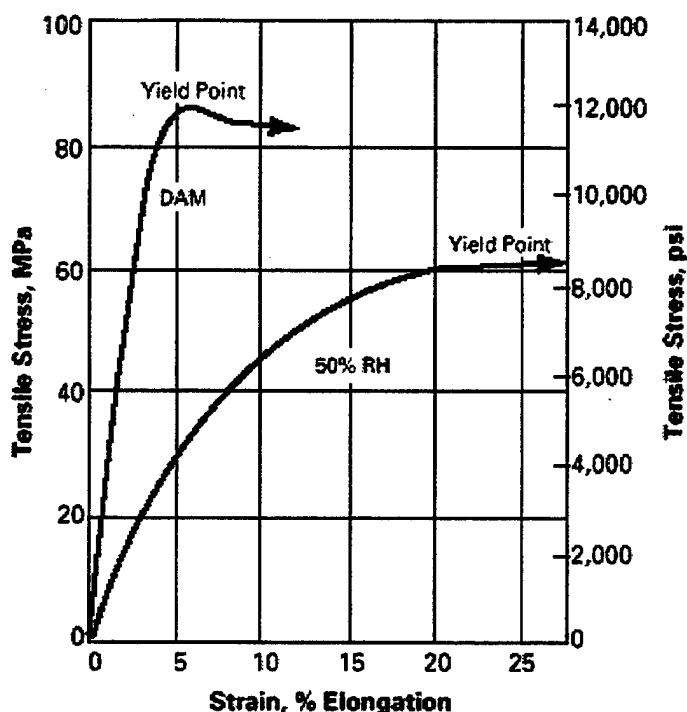


Figure 5. Tensile Stress-Strain Data for Nylon 6,6 at 23° C at 50% RH and Dry-as-Molded (DAM) Material Conditions (DuPont 1997).

Both increasing the temperature and increasing the moisture content reduce the stiffness and the yield point of the material. Figure 6 shows stress-vs.-strain curves for samples of nylon conditioned at 50% RH and four different temperature levels (DuPont 1997). At cold temperatures, the material displays brittle behavior, at higher temperatures, the material has tougher behavior. Figure 7 shows the effects of both moisture content and temperature on the flexural modulus of nylon 6,6 (DuPont 1997). Notice that over the normal operating temperature of the M865 (-25° F to 120° F), the modulus varies by a factor of 7, indicating that there can be substantial variation in obturator properties across the test temperature.

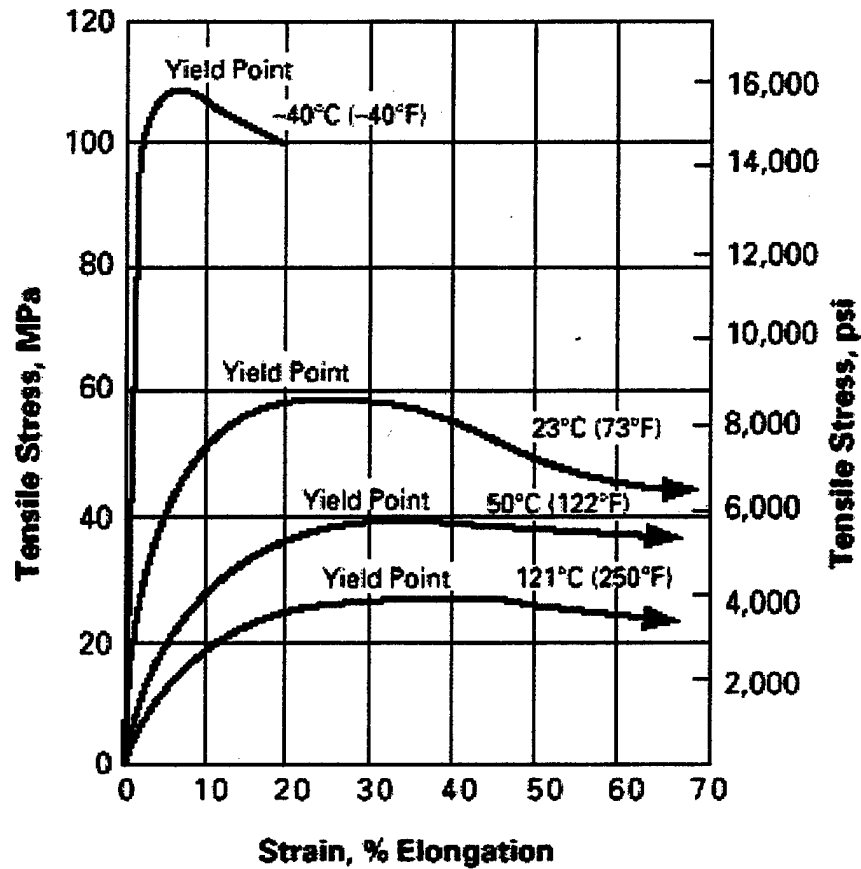


Figure 6. Tensile Stress-Strain Data for Nylon 6,6 at 50% RH at Four Different Temperatures (DuPont 1997).

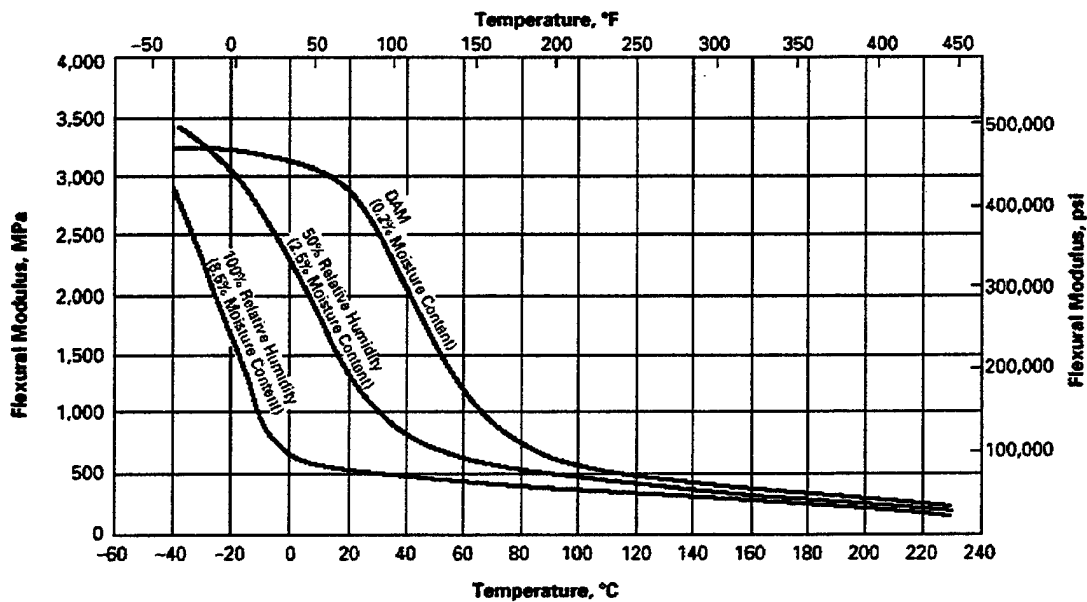


Figure 7. Flexural Modulus of Nylon 6,6 vs. Temperature at Various Moisture Contents (DuPont 1997).

Figure 8 and Figure 9 show the effects of moisture, temperature, and strain rate on the yield strength and elastic modulus of nylon 6,6. Moisture and temperature effects cause much greater changes in material properties than changes in the material strain rate.

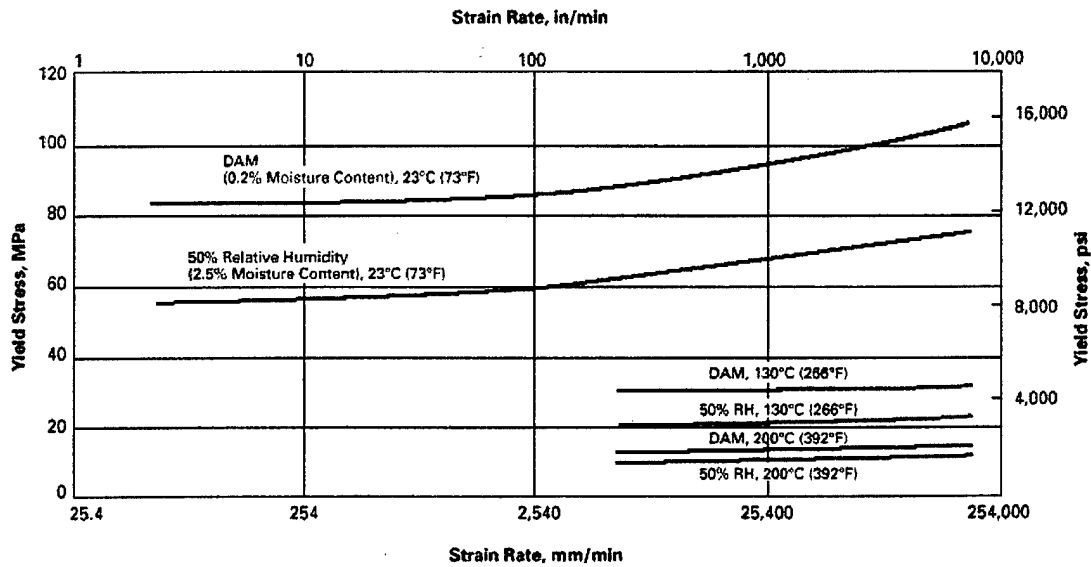


Figure 8. Yield Stress Data for Nylon 6,6 Dry-as-Molded And 50% RH vs. Strain Rate and Temperature (DuPont 1997).

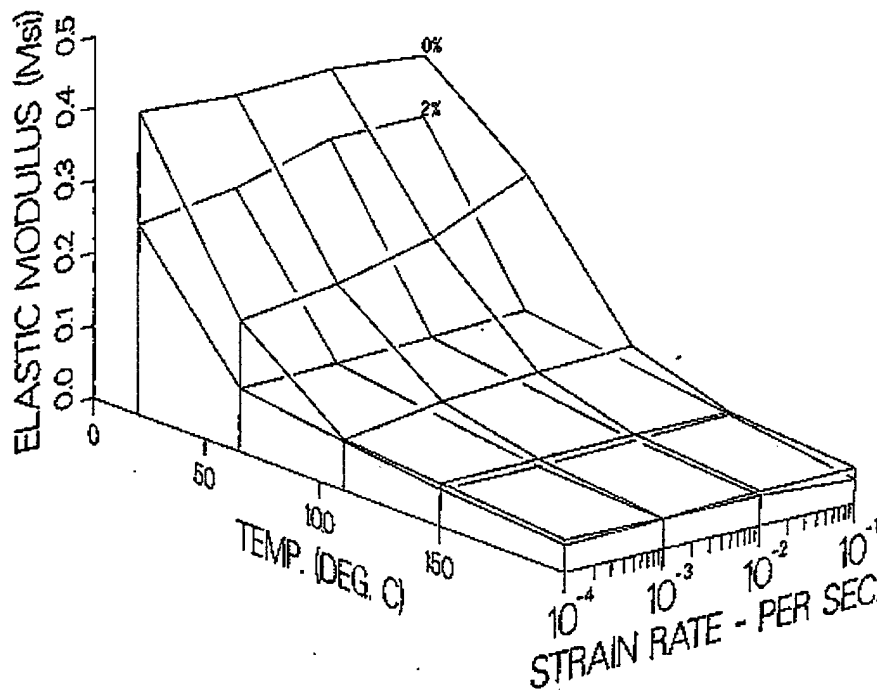


Figure 9. Effect of Temperature and Strain Rate on the Elastic Modulus of Nylon 6,6 at Two Moisture Levels (Kawahara, Brandon, and Korellis 1988).

2.4 Stress Concentration Due to Notch and Geometry.

While environmental effects can cause significant variation in the mechanical properties of the nylon obturator, the notches between sabot petals reduce variability in the behavior of the obturator during discard. In a separate study (Newill et al., to be published), the in-bore and discard behavior of the M865 projectile was numerically modeled. Figure 10 shows a finite element model showing the sabot discarding from the projectile. For simplicity of analysis, the notches in the obturator were not modeled. However, the analysis showed that during discard, the sabot geometry focused the stress in the obturator band such that the stress was three times higher at the sabot splits than in the surrounding material as shown in Figure 11. The stress is focused in a very small area because the obturator cannot slip on the knurled surface (due to the mechanical coupling) of the aluminum sabot. During discard, the elongation that occurs in the obturator will occur between the sabot petals. The distance between the sabot petals is very small, which in turn implies that the stress in the band is over a very short gauge length. This mechanism focuses all the energy from the petals separating into this very small area in the band, causing a large stress concentration.

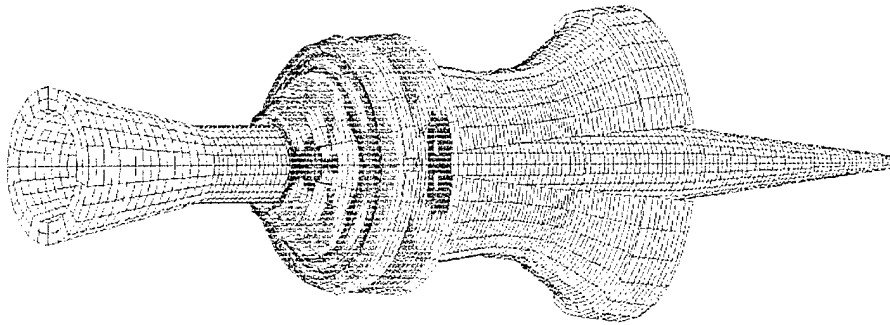


Figure 10. Finite Element Model of Sabot Discard.

The notch in the obturator further focuses the stress. The stress concentration due to the notch is defined by equation 1 (Hertzberg 1989):

$$k_t = \sqrt{\frac{a}{\rho}}, \quad (1)$$

where k_t is the stress concentration factor, a is the notch length, and ρ is the radius of the notch tip. For the notch lengths in the M865 (between 4 mm and 6 mm) with a notch radius of 0.25 mm, the stress concentration factor varies between 8 and 10.

The stress at the notch tip, due to a combination of the stresses from the sabot petals coming apart and the stress concentration at the notch, is then 24 to 30 times higher than the stress in the surrounding material. Since the stress at the notch tip is much higher than the failure strength of the nylon, variations in nylon material properties do not significantly affect failure of the band during discard.

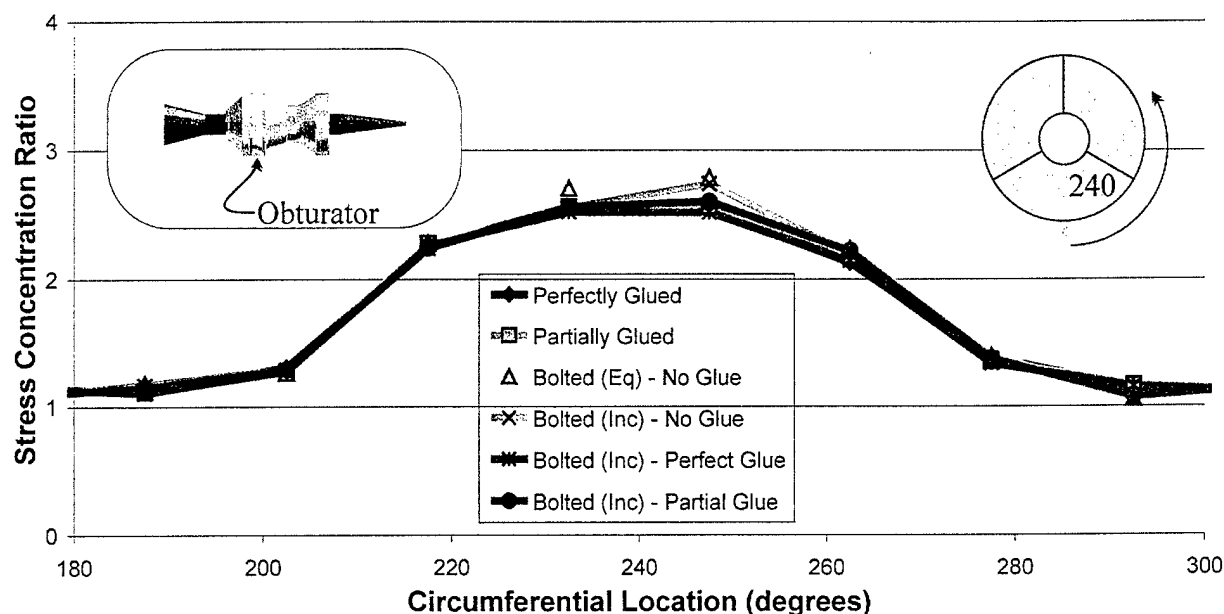


Figure 11. Circumferential Stress in the Obturator During Discard for an Unnotched Obturator.

3. Sealing Cuff

The investigation into the sealing cuff was more limited than the obturator portion. This was due to the development of the M865E3 version of the projectile, which is replacing the current sealing cuff with a nylon 6 snap ring adapter. In addition, material variations in the rubber have not been identified as a significant area of concern. A static break test was conducted on one projectile at Aberdeen Proving Ground (APG). In the test, the obturator dominated the lift-off process. Once the obturator fractured, the sealing cuff provided little resistance to the tearing lift-off loads. In this case, the RTV sealing cuff was well adhered to the sabot.

In the numerical discard analysis (Newill et al., to be published), the sealing cuff was modeled with several different interfacial conditions: perfectly bonded, partially bonded, no bond, perfectly bonded with a bolted sealing cuff, partially bonded with a bolted sealing cuff, and no bond

with a bolted sealing cuff. The stress distributions at the leading edge of the sealing cuffs during discard for these cases are shown in Figure 12 and Figure 13. When the sealing cuff is perfectly bonded to the sabot, the stress is focused at the seam between the petals. This provides a short gauge section for failure between the petals. When the sealing cuff is only partially bonded or not bonded, the circumferential stress is no longer focused between petals and a much larger section of the sealing cuff can deform prior to failure. Since the sealing cuff is an elastomer, it can endure substantial deformation and absorb significant energy before it breaks, increasing the chance for irregular failure. Therefore, a poor bond between the sabot and the sealing cuff can lead to less repeatable discard behavior, inducing variability that may contribute to poor Target Impact Dispersion (TID). When the sealing cuff is bolted to the sabot, the failure of the sealing cuff/sabot bond is less dramatic. The bolts act as secondary stress concentration sites, initiating failure if the adhesive fails. The bolts therefore help reduce the potential variability due to poor bonding.

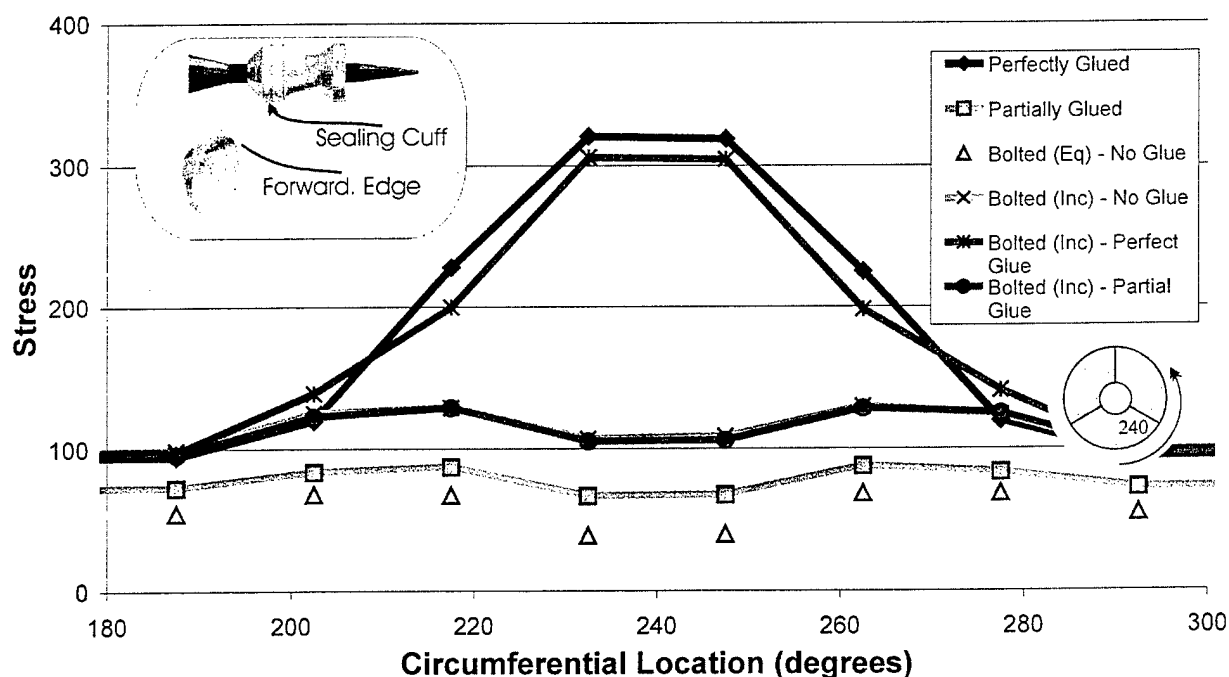


Figure 12. Stress Distribution at the Forward Edge of the Sealing Cuff During Sabot Discard.

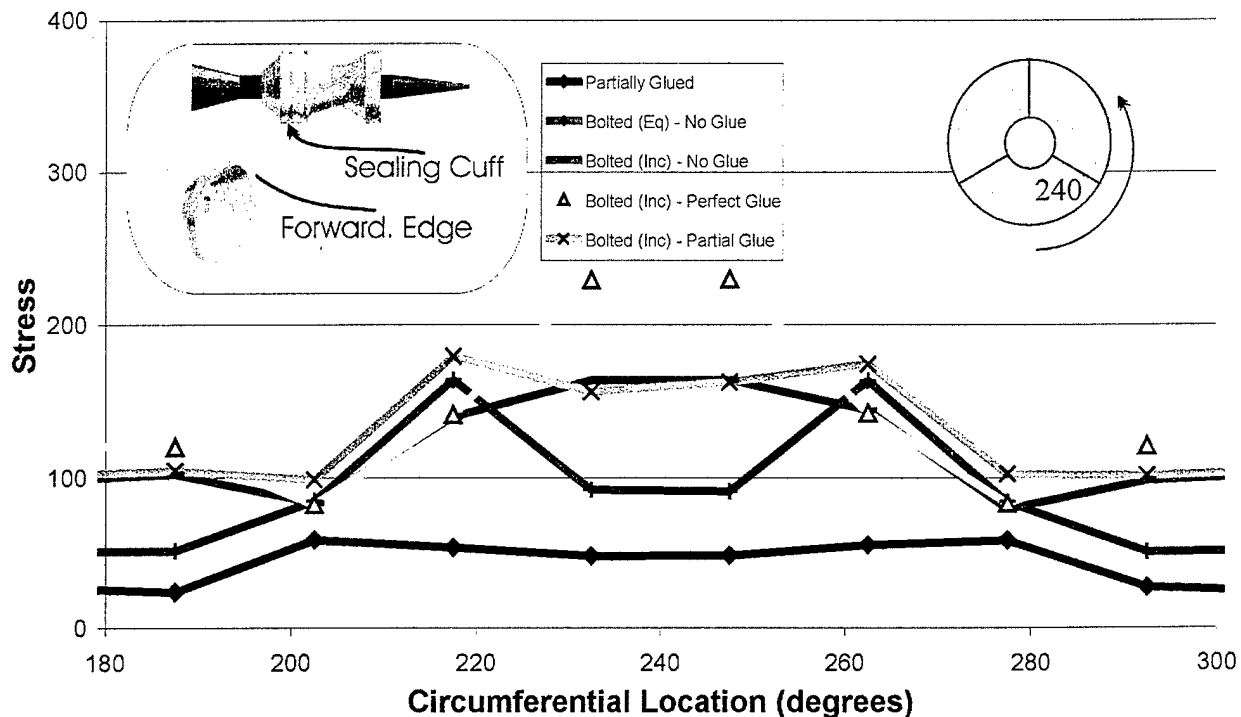


Figure 13. Stress Distribution at the Bolts in the Sealing Cuff During Sabot Discard.

4. Ballistic Testing

It was decided to pursue an experimental program to evaluate the effects of obturator properties on projectile performance. The two most important parameters that could affect obturator performance were identified as the toughness of the molded nylon band and the moisture content of the obturator. Three material conditions were then chosen for testing: a "brittle" condition with a low moisture content, a "tough" condition with a low moisture content, and a "tough" condition with a high moisture content (which further increases the toughness). The tough and brittle conditions were chosen based on reasonable molding conditions for the obturator and are described in section 4.1. The dry and wet environmental conditions were based on typical amounts of moisture in the obturator as described in section 4.2.

Based on an analysis prior to the test, it was determined that nine projectiles would need to be shot with each configuration to produce statistically meaningful results (Soencksen, Newill, and Webb, to be published). The ballistic test was also designed to evaluate the effects of bourrelet diameter on performance, which increased the number of configurations of test projectiles to

include two bourrelet diameters. Therefore, 60 projectiles (9 test projectiles for each configuration and 6 spare projectiles) were manufactured for the test. The test matrix is given in Table 1.

Table 1. Test Matrix (Number of Projectiles for Each Configuration)

Diameter (mm)	Dry, Brittle Obturator	Dry, Tough Obturator	Wet, Tough Obturator
119.69	9	9	9
119.83	9	9	9

4.1 Material Mechanical Properties.

Test obturators were manufactured in the two conditions “brittle” and “tough.” These conditions were achieved by controlling the processing parameters during the injection molding process. The details of the manufacture are contractor proprietary and therefore are not presented here. Several test specimens were manufactured with the same conditions, and their average mechanical properties are listed in Table 2. The tough specimens had approximately 4 times the elongation to failure as the brittle specimens.

Table 2. Mechanical Properties of the Molded Test Projectiles

Condition	Number of Test Specimens	Maximum Tensile Strength (psi)	Elastic Modulus (ksi)	Elongation to Failure (%)
Brittle	9	10487.6	191.2	11.79
Tough	6	9210.6	155.1	42.39

4.2 Environmental Conditions.

For the two environmental conditions (“dry” and “wet”), it was important to determine reasonable moisture levels for the projectiles (i.e., moisture contents that could be achieved in fielded ammunition). This would avoid biasing the test with “worst-case” environmental conditions such as an obturator completely saturated with moisture. Therefore, a study, described in section 4.2.1, was initiated to determine achievable moisture levels for nylon obturators. A second study, described in section 4.2.2, was then started to determine the best way to achieve these moisture levels.

4.2.1 Dry Out Testing.

The purpose of this test is to determine the moisture content of M865 projectiles that have been stored for long periods of time. Eight obturators were selected for testing. Four were manufactured in 1988 (lot number IOP88J058-003), and four were manufactured in 1997 (lot number MHM97K-002S295). One of the 1997 obturators broke in half when it was removed from the projectile, and each piece was used as a separate test specimen, so there were a total of nine test specimens. The 1988 projectiles had been stored in the open (not stored in ammunition cases) in bunkers at APG for approximately 10 years. The 1997 projectiles had been subjected to rough handling tests in December 1997, then sat for about 5 months in a propellant loading plant that had controlled temperature and humidity levels. The rough handling testing may have partially dried the obturators on these projectiles since it incorporates temperature cycling in a dry environment.

The projectiles were dried in an oven at 165° F for 32 days. The percent moisture loss vs. time is shown in Figure 14. The specimens from 1988 showed an average of 3.52% moisture loss by weight; the specimens from 1997 showed an average of 1.56% moisture loss by weight. Based on this study, it was determined that a reasonable moisture level for the “wet” obturators was 3.5% by weight moisture.

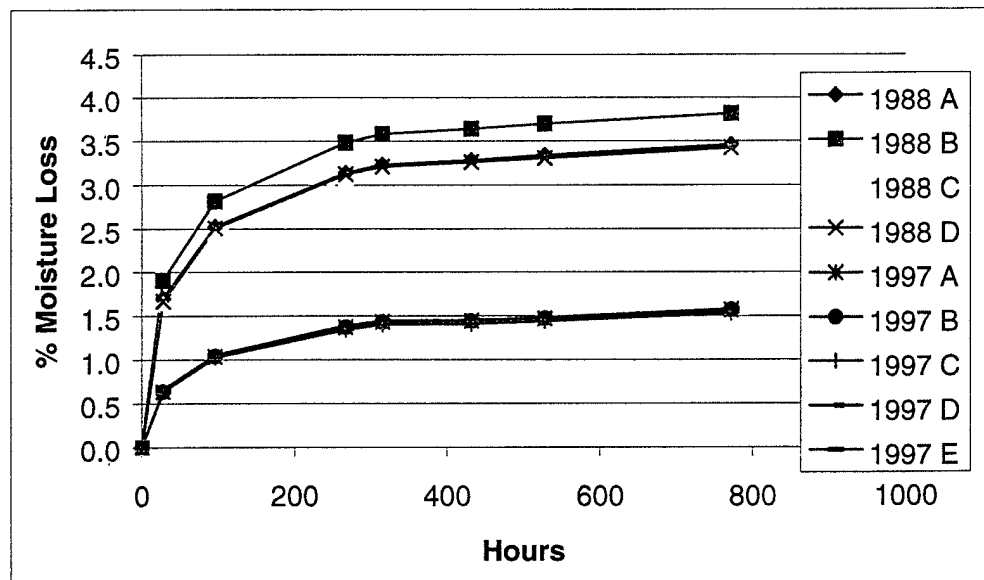


Figure 14. Obturator Moisture Loss vs. Time.

4.2.2 Moisture Absorption by the Obturators.

Moisture absorption tests were then initiated to determine the moisture saturation level and diffusion constants on the obturators. The obturators from the dry-out study were placed in two humidity chambers (50% RH and 90% RH) at 145° F. The percent weight gain vs. time is shown in Figure 15. The specimens conditioned at 90% RH had an average moisture saturation level of 5.67%. The specimens conditioned at 50% RH had an average saturation level of 1.93%.

From this study, it was interpolated that obturators with 3.5% moisture content would be in equilibrium in 75% RH air. It also showed that at 145° F, the obturators reached equilibrium moisture content within 20 days.

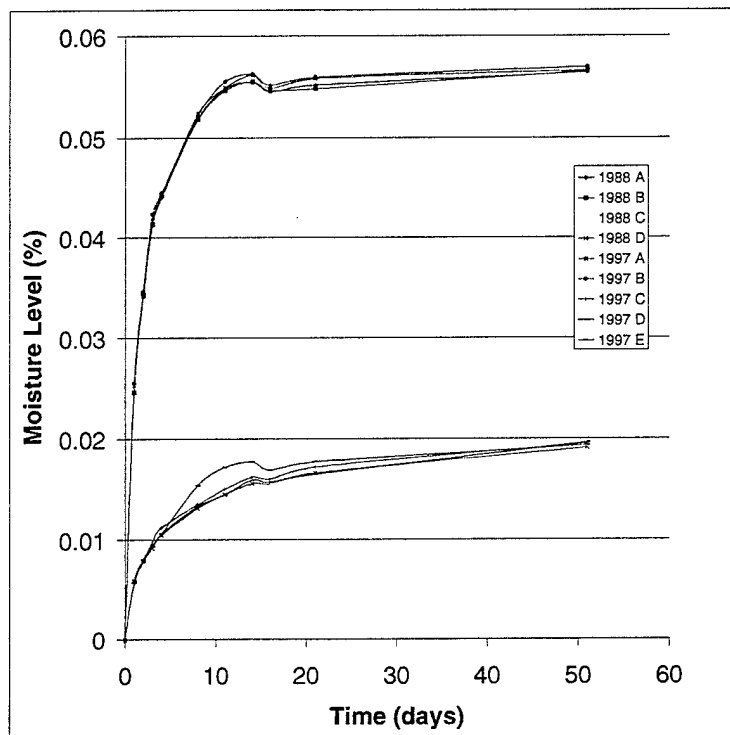


Figure 15. Percent Weight Gain vs. Time for Obturators Conditioned at 50% RH (Specimens 1988 C, 1988 D, 1997 C, 1997 D, and 1997 E) and 90% RH (Specimens 1988 A, 1988 B, 1997 A, and 1997 B).

4.2.3 Test Sample Preparation.

The test projectiles were manufactured in July of 1998. Three projectiles with removable obturators were made with the test projectiles in order to evaluate moisture content through the conditioning cycle. All of the projectiles were placed in ammunition storage cans and shipped to a

separate location for environmental conditioning. At that time, all of the "dry" projectiles were sealed in Mylar vacuum bags. The "wet" projectiles were placed into a conditioning chamber at 145° F at 95% RH until the test obturators showed a weight gain of 3.5%. After environmental conditioning, mechanical tests were performed on companion samples of all of the materials. The results are listed in Table 3. It should be noted that the maximum strength varies by a factor of 2, the elastic modulus varies by a factor of 25, and the elongation to failure varies by a factor 10 for the obturators.

Table 3. Average Obturator Mechanical Properties After Environmental Conditioning

Condition	Number of Test Specimens	Maximum Tensile Strength (psi)	Elastic Modulus (ksi)	Elongation to Failure (%)
Brittle	23	11057.9	576.0	7.75
Tough	12	9569.9	380.87	39.65
Tough-Wet	15	5907.7	23.4	71.62

The "wet" projectiles were then placed in Mylar vacuum bags, and all of the projectiles were shipped to an ammunition loading plant. At the load plant, the vacuum bags were removed and the projectiles were loaded, placed in ammunition storage cans, and shipped to the ARL Transonic Experimental Facility at APG, MD, for testing.

At the Transonic Experimental Facility, the projectiles were temperature-conditioned in environmental chambers prior to the test. The "dry" projectiles were stored at 120° F and 25% RH for a minimum of 24 hours prior to testing. The "wet projectiles" were conditioned at 120° F and 75% RH for a minimum of 72 hours and a maximum of 120 hours prior to testing. The dummy obturators were weighed prior to the ballistic test, and they had an average moisture content of 3.35%. The reason for conditioning the "wet" projectile with humidity for longer periods of times was twofold. First, since the testing was fired with a propellant temperature of 120° F, the moisture content of the obturators would have dropped due to drying. The conditions were chosen to bring the obturators back to the 3.5% moisture content. While the timeframe was too short to fully recondition the obturators, the critical area of the band is the base of the slot since this is where the failure will initiate. The condition just before firing will ensure that this region is at the appropriate moisture content.

4.3 Results.

The full results of the ballistic test will be described in a separate report (Soencksen, Newill, and Webb, to be published). However, this section reviews the results significant to the obturator performance. The projectiles were fired from an M1A1 tank, through the Transonic Experimental Facility. Spark shadowgraphs were used to establish the yawing motion parameters, which were extrapolated to determine first max yaw. Target impact was also recorded for each shot. The test was conducted on the E3 version of the M865 projectile as shown in Figure 16. The E3 version of the M865 differs from the original version in that it incorporates a nylon 6 snap ring adapter rather than the rubber sealing cuff used on the original version of the M865 as shown in Figure 17. The variability in the first max yaw results from the ballistic testing as measured through standard deviations was 0.44 for the dry brittle bands, 0.60 for the dry tough bands, and 0.47 for the "wet" tough bands (Soencksen, Newill, and Webb, to be published).

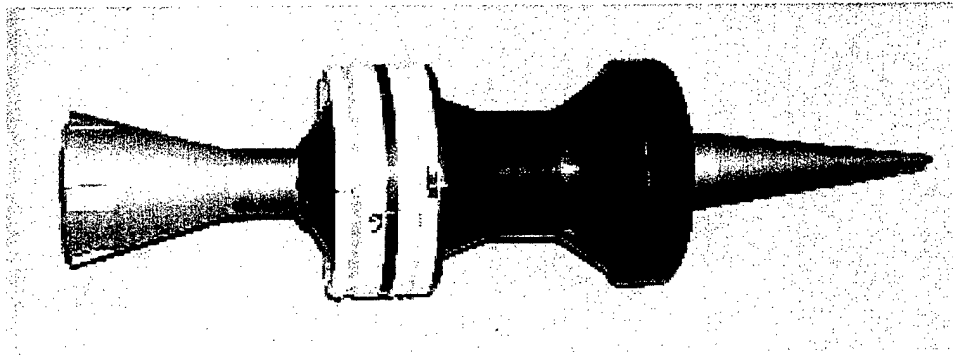


Figure 16. E3 Version of the M865 Projectile.

During the test, anomalies were noted in the fracture of the obturator. Several large pieces of obturators were recovered on the pad in front of the tank. The length of many of the pieces found was greater than that of the 120° sabot segment arc, which would have been expected for normal band breakage. Several of these pieces are shown in Figure 18 and Figure 19. The bands also showed signs of gas leakage underneath the obturators (Figure 20), and the aft portion of the bands were missing or badly damaged. The remaining sections of the aft portion of the band had a triangular cross section, which implies that they were worn irregularly due to gas leakage underneath pressing the band against the tube. Since gas leaked underneath the aft portion of the band in-bore, the loss of tube support at muzzle exit caused the aft portion of the band to blow off of the projectile. The leakage underneath the obturator and loss of the aft portion of the band

disengages the knurled surface on the band seat. When the knurled surfaces are not engaged, the stresses in the band are not focused between the sabot petals during discard, leading to more erratic fracture. This is further supported with the recovery of the large section of obturator from the testing. The bands show that they did not fail at each of the slots as designed. Since the bands are not fracturing as designed, the effects of the obturator mechanical properties should be more evident. This is due to the reduction in stress concentration (predicted in section 2) due to the loss of the mechanical coupling from the knurling surface. It also allows the obturator to absorb energy over a larger area, leading to failure that is more erratic. However, the results show that there were no significant differences in obturator behavior, indicating that even with the reduced stress concentration, the differences in mechanical property still did not significantly affect discard. This also implies that if the band is performing properly (with the knurling surface intact), the material differences should have even less effect.



Figure 17. Comparison of the Original M865 Projectile (Left) to the E3 Version (Right).

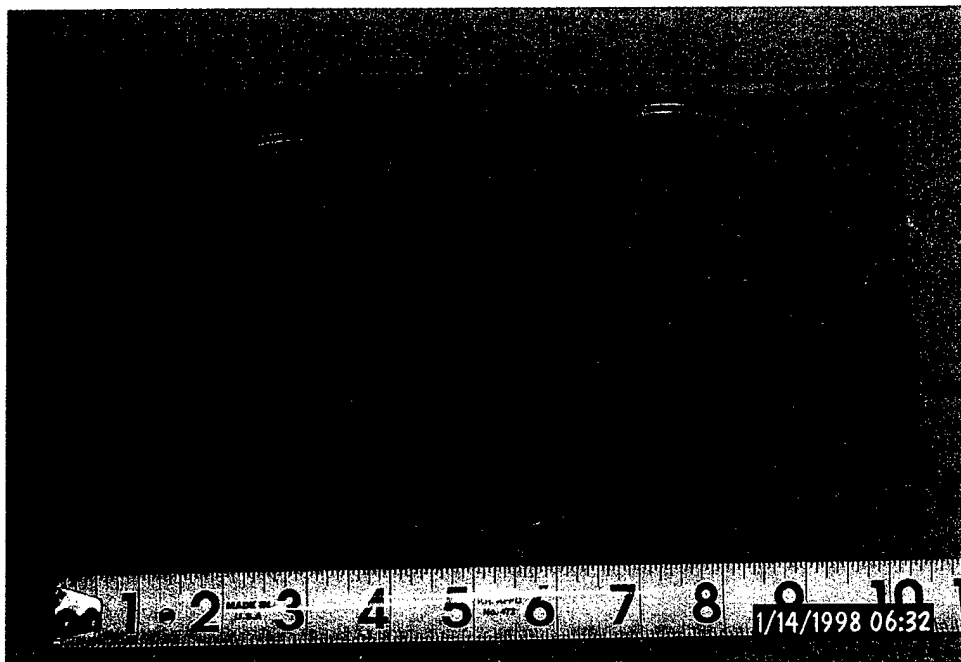


Figure 18. Obturator Pieces Found During the Ballistic Test.

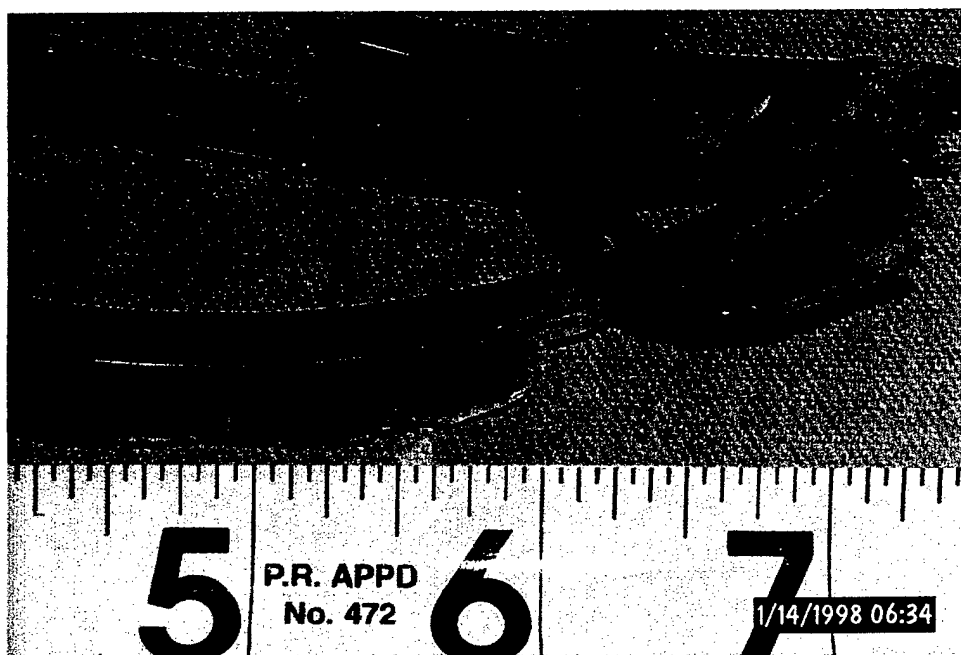


Figure 19. Close-up Photograph of Obturator Pieces Found During the Ballistic Test.



Figure 20. Sabot From a Separate Ballistic Test Showing Soot in the Obturator Seat Due to Gas Leakage Underneath Obturator.

5. Conclusions

Historical test data have shown that nylon functions well as an obturator material. It is used in many different types of ammunition and rarely causes problems. However, nylon can have a variety of properties, and these need to be understood. The toughness of the obturator can change significantly due to processing conditions, moisture absorption, and temperature. In the M865 obturator, variability due to processing is offset by the knurled geometry and notch, which focus the stress at the sabot seams. The ballistic tests in this study confirm that variability in the mechanical properties of nylon has little influence on sabot discard.

The obturator has several functions, which are contradictory with regard to the material requirements. During the manufacturing, handling, and storage of the projectiles, the obturator needs to be tough to avoid brittle cracking although nylon is most brittle in its dry-as-molded condition. As the obturator is exposed to ambient humidity levels, it will absorb moisture and increase its toughness. During discard, the obturator needs to fail in a consistent manner for each shot.

As described in section 2, the mechanical properties of nylon can vary by several orders of magnitude due to the processing conditions, moisture content, and temperature. This means that the obturator fracture can vary significantly due to the material properties. Therefore, an engineered

breaking mechanism was designed into the obturator and obturator seat to overcome the material property variability. The knurled surface in the obturator seat on the sabot and the notch in the forward edge of the obturator both help to focus the stress and achieve repeatable failure. These failure mechanisms minimize the variability due the nylon mechanical properties during discard and, therefore, minimize the potential shot-to-shot variability.

The obturators used on the projectile in ballistic testing were made with a variety of material conditions. The results in Table 3 show that the maximum strength varied by a factor of 2, the elastic modulus varied by a factor of 25, and the elongation to failure varied by a factor 10 for the obturators in this study. It should be noted that these ranges of material properties do not represent extremes mechanical properties for nylon; rather, they are all conditions that could be reasonably seen in tank ammunition. However, the ballistic test showed no significant difference in first max yaw of the projectile's behavior, indicating that the variability in nylon behavior can be overcome with an engineered failure mechanism and therefore had little influence on projectile discard even with leakage problems underneath the obturator.

An issue that needs to be monitored is the manufacturing conditions of the obturators since these impact the material properties. Currently, there are no quality control tests or acceptance criteria for the molded projectiles, allowing the crystallinity and void content to vary significantly. While the ballistic testing showed that the variability can be overcome with mechanical fracture mechanism, controlling the source of the variability will ensure more consistent obturator performance.

The most significant issue with the sealing cuff appears to be adhesion to the sabot. Good adhesion focuses the circumferential stress in the sealing cuff at the seams and leads to consistent fracture. If the adhesion is poor or the adhesive interface fails, the bolts act as secondary fracture initiation sites. While this acts as an engineered failure mechanism, it is not as well done as the knurling/slot failure mechanism in the obturator. It appears that this portion is working well enough due to the good TID performance of the projectiles.

6. References

- American Society for Testing and Materials. *Standard Classification System for Nylon Injection and Extrusion Materials (PA)*. ASTM D 4066-96a, West Conshohocken, PA, 1996.
- DuPont Engineered Materials. *Zytel/Minlon Design Guide-Module II*. www.DuPont.com, 1997.
- Hertzberg, R. W. *Deformation and Fracture Mechanics of Engineering Materials*. New York: John Wiley & Sons, 1989.
- Kawahara, W. A., S. L. Brandon, and J. S. Korellis. "Temperature, Moisture, and Strain Rate Effects on the Compressive Mechanical Behavior of Nylon 6/6." Sandia National Laboratory, Albuquerque, NM, 1988.
- Kohan, M. I. *Nylon Plastics*. New York: John Wiley & Sons, 1973.
- Newill, J. F., M.S. Berman, J. Despirito, K.P. Soencksen, and C. P. R. Hoppel. "Simulation of M865 and M865E3 Launch Characteristics with Comparison to Ballistic Testing." Technical Report, U.S. Army Research Laboratory, Aberdeen Proving Ground, MD, in progress.
- Soencksen, K. P., J. F. Newill, and D. W. Webb "The Effect of Certain Process Variables on the Launch and Flight of the M865E3 Projectile." Technical Report, U.S. Army Research Laboratory, Aberdeen Proving Ground, MD, in progress.
- Tsai, S. W. *Composites Design*. 4th Edition, Dayton, OH: Think Composites, 1988.

INTENTIONALLY LEFT BLANK.

Bibliography

- Galeski, A. "Morphology of Bulk Nylon 6 Subjected to Plane Strain Compression." *Macromolecules*, vol. 24, no. 13, pp. 3953-3961, 1991.
- Kawahara, W. A., J. T. Totten, and J.S. Korellis. "Effects of Temperature and Strain Rate on the Nonlinear Compressive Mechanical Behavior of Polypropylene." SAND89-8233, Sandia National Laboratories, Livermore, CA 1989.
- Newill, J. F., C. P. R. Hoppel, and M. S. Berman. "Coupled Modeling of Moisture and Temperature Effects to Predict Deformation." *American Society for Composites' Thirteenth Technical Conference*, pp. 1003-1013, Baltimore, MD, 1998.
- Newill, J. F., S. H. McKnight, C. P. R. Hoppel, G. R. Cooper, and M. S. Berman. "Theoretical Evaluation of Moisture Protection Using Coatings." *U.S. Army Symposium on Solid Mechanics Proceedings*, Myrtle Beach, SC, 1999.
- Newill, J. F., S. A. Wilkerson, C. P. R. Hoppel, and W. H. Drysdale. "Numerical Simulation of Composite Kinetic Energy Projectiles Launched by an M1A1 Abrams M256 Gun System." *30th SAMPE Technical Conference Proceedings*, San Antonio, TX, 1998.
- Smith, M. T., and E. M. Patton. "Evaluation of the Deformation Behavior of Nylon Materials Used in Ballistic Applications." BRL-CR-554, U.S. Army Ballistic Research Laboratory, Aberdeen Proving Ground, MD, 1986.

INTENTIONALLY LEFT BLANK.

NO. OF
COPIES ORGANIZATION

2 DEFENSE TECHNICAL
INFORMATION CENTER
DTIC DDA
8725 JOHN J KINGMAN RD
STE 0944
FT BELVOIR VA 22060-6218

1 HQDA
DAMO FDQ
D SCHMIDT
400 ARMY PENTAGON
WASHINGTON DC 20310-0460

1 OSD
OUSD(A&T)/ODDDR&E(R)
R J TREW
THE PENTAGON
WASHINGTON DC 20301-7100

1 DPTY CG FOR RDE HQ
US ARMY MATERIEL CMD
AMCRD
MG CALDWELL
5001 EISENHOWER AVE
ALEXANDRIA VA 22333-0001

1 INST FOR ADVNCD TCHNLGY
THE UNIV OF TEXAS AT AUSTIN
PO BOX 202797
AUSTIN TX 78720-2797

1 DARPA
B KASPAR
3701 N FAIRFAX DR
ARLINGTON VA 22203-1714

1 NAVAL SURFACE WARFARE CTR
CODE B07 J PENNELLA
17320 DAHLGREN RD
BLDG 1470 RM 1101
DAHLGREN VA 22448-5100

1 US MILITARY ACADEMY
MATH SCI CTR OF EXCELLENCE
DEPT OF MATHEMATICAL SCI
MAJ M D PHILLIPS
THAYER HALL
WEST POINT NY 10996-1786

NO. OF
COPIES ORGANIZATION

1 DIRECTOR
US ARMY RESEARCH LAB
AMSRL DD
J J ROCCHIO
2800 POWDER MILL RD
ADELPHI MD 20783-1145

1 DIRECTOR
US ARMY RESEARCH LAB
AMSRL CS AS (RECORDS MGMT)
2800 POWDER MILL RD
ADELPHI MD 20783-1145

3 DIRECTOR
US ARMY RESEARCH LAB
AMSRL CI LL
2800 POWDER MILL RD
ADELPHI MD 20783-1145

ABERDEEN PROVING GROUND

4 DIR USARL
AMSRL CI LP (305)

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
1	DIRECTOR USARL AMSRL CP CA D SNIDER 2800 POWDER MILL RD ADELPHI MD 20783-1145
1	COMMANDER US ARMY ARDEC AMSTA AR FSE T GORA PICATINNY ARSENAL NJ 07806-5000
3	COMMANDER US ARMY ARDEC AMSTA AR TD PICATINNY ARSENAL NJ 07806-5000
5	US ARMY TACOM AMSTA JSK S GOODMAN J FLORENCE AMSTA TR D B RAJU L HINOJOSA D OSTBERG WARREN MI 48397-5000
5	PM SADARM SFAE GCSS SD COL B ELLIS M DEVINE W DEMASSI J PRITCHARD S HROWNAK PICATINNY ARSENAL NJ 07806-5000
1	COMMANDER US ARMY ARDEC SFAE FAS PM F MCLAUGHLIN PICATINNY ARSENAL NJ 07806

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
4	COMMANDER US ARMY ARDEC AMSTA AR CCH S MUSALLI R CARR M LUCIANO T LOUCEIRO PICATINNY ARSENAL NJ 07806-5000
2	COMMANDER US ARMY ARDEC AMSTA AR E FENNELL PICATINNY ARSENAL NJ 07806-5000
1	COMMANDER US ARMY ARDEC AMSTA AR CCH PICATINNY ARSENAL NJ 07806-5000
2	COMMANDER US ARMY ARDEC AMSTA AR PICATINNY ARSENAL NJ 07806-5000
3	COMMANDER US ARMY ARDEC AMSTA AR CCH P J LUTZ AMSTA AR FSF T C LIVECCHIA AMSTA AR QAC T/C C PATEL PICATINNY ARSENAL NJ 07806-5000
2	COMMANDER US ARMY ARDEC AMSTA AR M D DEMELLA F DIORIO PICATINNY ARSENAL NJ 07806-5000

<u>NO. OF</u> <u>COPIES</u>	<u>ORGANIZATION</u>
3	COMMANDER US ARMY ARDEC AMSTA AR FSA A WARNASH B MACHAK M CHIEFA PICATINNY ARSENAL NJ 07806-5000
1	COMMANDER WATERVLIET ARSENAL SMCWV QAE Q B VANINA BLDG 44 WATERVLIET NY 12189-4050
1	COMMANDER WATERVLIET ARSENAL SMCWV SPM T MCCLOSKEY BLDG 253 WATERVLIET NY 12189-4050
8	DIRECTOR BENET LABORATORIES AMSTA AR CCB J KEANE J BATTAGLIA J VASILAKIS G FFIAR V MONTVORI G DANDREA R HASENBEIN SMCAR CCB R S SOPOK WATERVLIET NY 12189
1	COMMANDER WATERVLIET ARSENAL SMCWV QA QS K INSCO WATERVLIET NY 12189-4050
1	COMMANDER US ARMY ARDEC PRDCTN BASE MODERN ACTY AMSMC PBM K PICATINNY ARSENAL NJ 07806-5000
1	COMMANDER US ARMY BELVOIR RD&E CTR STRBE JBC FT BELVOIR VA 22060-5606

<u>NO. OF</u> <u>COPIES</u>	<u>ORGANIZATION</u>
2	COMMANDER US ARMY ARDEC AMSTA AR FSP G M SCHIKSNIS D CARLUCCI PICATINNY ARSENAL NJ 07806-5000
1	US ARMY COLD REGIONS RESEARCH & ENGINEERING LABORATORY P DUTTA 72 LYME RD HANOVER NH 03755
1	DIRECTOR USARL AMSRL WT L D WOODBURY 2800 POWDER MILL RD ADELPHI MD 20783-1145
3	COMMANDER US ARMY AMCOM AMSMI RD W MCCORKLE AMSMI RD ST P DOYLE AMSMI RD ST CN T VANDIVER REDSTONE ARSENAL AL 35898-5247
2	US ARMY RESEARCH OFFICE A CROWSON J CHANDRA PO BOX 12211 RESEARCH TRIANGLE PARK NC 27709-2211
3	US ARMY RESEARCH OFFICE ENGINEERING SCIENCES DIV R SINGLETON G ANDERSON K IYER PO BOX 12211 RESEARCH TRIANGLE PARK NC 27709-2211

NO. OF
COPIES ORGANIZATION

- 5 PROJECT MANAGER
TANK MAIN ARMAMENT SYSTEMS
SFAE GSSC TMA
COL PAWLICKI
K KIMKER
E KOPACZ
R ROESER
B DORCY
PICATINNY ARSENAL NJ
07806-5000
- 1 PROJECT MANAGER
TANK MAIN ARMAMENT SYS
SFAE GSSC TMA SMD
R KOWALSKI
PICATINNY ARSENAL NJ
07806-5000
- 2 PEO FIELD ARTILLERY SYSTEMS
SFAE FAS PM H GOLDMAN
T MCWILLIAMS
PICATINNY ARSENAL NJ
07806-5000
- 2 PROJECT MANAGER CRUSADER
G DELCOCO
J SHIELDS
PICATINNY ARSENAL NJ
07806-5000
- 2 NASA LANGLEY RESEARCH CTR
AMSRL VS MS 266
W ELBER
F BARTLETT JR
HAMPTON VA 23681-0001
- 2 COMMANDER
DARPA
J KELLY
B WILCOX
3701 N FAIRFAX DR
ARLINGTON VA 22203-1714

NO. OF
COPIES ORGANIZATION

- 6 COMMANDER
WRIGHT PATTERSON AFB
WL FIV A MAYER
WL MLBM
S DONALDSON
T BENSON-TOLLE
C BROWNING
J MCCOY
F ABRAHAMS
2941 P STREET STE 1
DAYTON OH 45433
- 1 NAVAL SURFACE WARFARE CTR
DAHLGREN DIV CODE G06
DAHLGREN VA 22448
- 1 NAVAL RESEARCH LABORATORY
I WOLOCK CODE 6383
WASHINGTON DC 20375-5000
- 1 OFFICE OF NAVAL RESEARCH
MECH DIV CODE 1132SM
Y RAJAPAKSE
ARLINGTON VA 22217
- 1 NAVAL SURFACE WARFARE CTR
CRANE DIVISION
M JOHNSON CODE 20H4
LOUISVILLE KY 40214-5245
- 1 DAVID TAYLOR RESEARCH CTR
SHIP STRUCTURES & PROTECTION
DEPARTMENT
J CORRADO CODE 1702
BETHESDA MD 20084
- 2 DAVID TAYLOR RESEARCH CTR
R ROCKWELL
W PHYLLAIER
BETHESDA MD 20054-5000
- 1 DEFENSE NUCLEAR AGENCY
INNOVATIVE CONCEPTS DIV
R ROHR
6801 TELEGRAPH RD
ALEXANDRIA VA 22310-3398

NO. OF
COPIES ORGANIZATION

1 EXPEDITIONARY WARFARE DIV
F SHOUP N85
2000 NAVY PENTAGON
WASHINGTON DC 20350-2000

1 OFFICE OF NAVAL RESEARCH
D SIEGEL 351
800 N QUINCY ST
ARLINGTON VA 22217-5660

1 NAVAL SURFACE WARFARE CTR
J H FRANCIS CODE G30
DAHLGREN VA 22448

2 NAVAL SURFACE WARFARE CTR
D WILSON CODE G32
R D COOPER CODE G32
DAHLGREN VA 22448

4 NAVAL SURFACE WARFARE CTR
J FRAYSSE CODE G33
E ROWE CODE G33
T DURAN CODE G33
L DE SIMONE CODE G33
DAHLGREN VA 22448

1 COMMANDER
NAVAL SEA SYSTEMS CMD
D LIESE
2531 JEFFERSON DAVIS HIGHWAY
ARLINGTON VA 22242-5160

1 NAVAL SURFACE WARFARE CTR
M E LACY CODE B02
17320 DAHLGREN RD
DAHLGREN VA 22448

1 NAVAL SURFACE WARFARE CTR
TECH LIBRARY CODE 323
17320 DAHLGREN RD
DAHLGREN VA 22448

4 DIRECTOR
LLNL
R CHRISTENSEN
S DETERESA
F MAGNESS
M FINGER
PO BOX 808
LIVERMORE CA 94550

NO. OF
COPIES ORGANIZATION

1 LOS ALAMOS NATL LAB
F ADDESSIO MS B216
PO BOX 1633
LOS ALAMOS NM 87545

1 LOS ALAMOS NATL LAB
J REPPA MS F668
PO BOX 1663
LOS ALAMOS NM 87545

1 OAK RIDGE NATIONAL LABORATORY
R M DAVIS
PO BOX 2008
OAK RIDGE TN 37831-6195

1 PENNSYLVANIA STATE UNIVERSITY
C BAKIS
227 N HAMMOND
UNIVERSITY PARK PA 16802

3 UNITED DEFENSE LP
4800 EAST RIVER RD
P JANKE MS170
T GIOVANETTI MS236
B VAN WYK MS389
MINNEAPOLIS MN 55421-1498

4 DIRECTOR
SANDIA NATL LABORATORIES
APPLIED MECHANICS DEPT
DIVISION 8241
W KAWAHARA
K PERANO
D DAWSON
P NIELAN
PO BOX 969
LIVERMORE CA 94550-0096

1 DREXEL UNIVERSITY
A S D WANG
32ND AND CHESTNUT ST
PHILADELPHIA PA 19104

1 BATTELLE
C R HARGREAVES
505 KING AVE
COLUMBUS OH 43201-2681

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
1	PACIFIC NORTHWEST LABORATORY M SMITH PO BOX 999 RICHLAND WA 99352
1	LLNL M MURPHY PO BOX 808 L 282 LIVERMORE CA 94550
1	NORTH CAROLINA STATE UNIVERSITY CIVIL ENGINEERING DEPT W RASDORF PO BOX 7908 RALEIGH NC 27696-7908
1	PENNSYLVANIA STATE UNIVERSITY R MCNITT 227 HAMMOND BLDG UNIVERSITY PARK PA 16802
1	PENNSYLVANIA STATE UNIVERSITY R S ENGEL 245 HAMMOND BLDG UNIVERSITY PARK PA 16801
1	PURDUE UNIVERSITY SCHOOL OF AERO & ASTRO C T SUN W LAFAYETTE IN 47907-1282
1	STANFORD UNIVERSITY DEPT OF AERONAUTICS AND AEROBALLISTICS DURANT BUILDING S TSAI STANFORD CA 94305
1	UCLA MANE DEPT ENGR IV H THOMAS HAHN LOS ANGELES CA 90024-1597

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
2	U OF DAYTON RSCH INSTITUTE R Y KIM A K ROY 300 COLLEGE PARK AVE DAYTON OH 45469-0168
1	UNIVERSITY OF DAYTON J M WHITNEY COLLEGE PARK AVE DAYTON OH 45469-0240
2	UNIVERSITY OF DELAWARE CTR FOR COMPOSITE MATERIALS J GILLESPIE M SANTARE 201 SPENCER LABORATORY NEWARK DE 19716
1	UNIV OF ILLINOIS AT URBANA CHAMPAIGN NATL CTR FOR COMPOSITE MATERIALS RESEARCH 216 TALBOT LABORATORY J ECONOMY 104 S WRIGHT STREET URBANA IL 61801
1	UNIVERSITY OF KENTUCKY L PENN 763 ANDERSON HALL LEXINGTON KY 40506-0046
1	UNIVERSITY OF UTAH DEPT OF MECH & INDUSTRIAL ENGR S SWANSON SALT LAKE CITY UT 84112
2	UNIV OF TEXAS AT AUSTIN CTR FOR ELECTROMECHANICS A WALLS J KITZMILLER 10100 BURNET RD AUSTIN TX 78758-4497

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
3	VA POLYTECHNICAL INSTITUTE & STATE UNIVERSITY DEPT OF ESM M W HYER K REIFSNIDER R JONES BLACKSBURG VA 24061-0219
1	UNIVERSITY OF MARYLAND DEPT OF AEROSPACE ENGR A J VIZZINI COLLEGE PARK MD 20742
1	AAI CORPORATION T G STASTNY PO BOX 126 HUNT VALLEY MD 21030-0126
1	JOHN HEBERT G CHRYSSOMALLIS PO BOX 1072 HUNT VALLEY MD 21030-0126
1	ARMTEC DEFENSE PRODUCTS S DYER 85 901 AVE 53 PO BOX 848 COACHELLA CA 92236
2	ADVANCED COMPOSITE MATERIALS CORPORATION P HOOD J RHODES 1525 S BUNCOMBE RD GREER SC 29651-9208
1	SAIC D DAKIN 2200 POWELL ST STE 1090 EMERYVILLE CA 94608
1	SAIC M PALMER 2109 AIR PARK RD S E ALBUQUERQUE NM 87106
1	SAIC R ACEBAL 1225 JOHNSON FERRY RD STE 100 MARIETTA GA 30068

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
1	SAIC G CHRYSSOMALLIS 3800 W 80TH STREET STE 1090 BLOOMINGTON MN 55431
6	ALLIANT TECH SYSTEMS INC C CANDLAND R BECKER L LEE C AACHUS D KAMDAR D FISHER 600 2ND ST NE HOPKINS MN 55343-8367
1	AMOCO PERFORMANCE PRODUCTS INC M MICHNO JR 4500 MCGINNIS FERRY RD ALPHARETTA GA 30202-3944
1	APPLIED COMPOSITES W GRISCH 333 NORTH SIXTH ST ST CHARLES IL 60174
1	BRUNSWICK DEFENSE T HARRIS STE 410 1745 JEFFERSON DAVIS HWY ARLINGTON VA 22202
1	PROJECTILE TECHNOLOGY INC 515 GILES ST HAVRE DE GRACE MD 21078
1	CUSTOM ANALYTICAL ENGR SYS INC A ALEXANDER 13000 TENSOR LANE NE FLINTSTONE MD 21530
1	NOESIS INC A BOUTZ 1110 N GLEBE RD STE 250 ARLINGTON VA 22201-4795

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
1	ARROW TECH ASSO 1233 SHELBURNE RD STE D 8 SOUTH BURLINGTON VT 05403-7700
1	NAVAL SURFACE WARFARE CTR R HUBBARD G33-C DAHLGREN DIVISION DAHLGREN VA 2248-5000
5	GEN CORP AEROJET D PILLASCH T COULTER C FLYNN D RUBAREZUL M GREINER 1100 WEST HOLLYVALE ST AZUSA CA 91702-0296
7	CIVIL ENGR RSCH FOUNDATION H BERNSTEIN PRESIDENT C MAGNELL K ALMOND R BELLE M WILLETT E DELO B MATTES 1015 15TH ST NW STE 600 WASHINGTON DC 20005
1	NATIONAL INSTITUTE OF STANDARDS AND TECHNOLOGY STRUCTURE & MECHANICS GP POLYMER DIV POLYMERS G MCKENNA RM A209 GAITHERSBURG MD 20899
1	DUPONT COMPANY COMPOSITES ARAMID FIBERS S BORLESKE DEVELOPMENT MGR CHESNUT RUN PLAZA PO BOX 80702 WILMINGTON DE 19880-0702
1	GENERAL DYNAMICS LAND SYSTEMS DIVISION D BARTLE PO BOX 1901 WARREN MI 48090

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
3	HERCULES INC R BOE F POLICELLI J POESCH PO BOX 98 MAGNA UT 84044
3	HERCULES INC G KUEBELER J VERMEYCHUK B MANDERVILLE JR HERCULES PLZ WILMINGTON DE 19894
1	HEXCEL M SHELENDICH 11555 DUBLIN BLVD PO BOX 2312 DUBLIN CA 94568-0705
4	INSTITUTE FOR ADVANCED TECH H FAIR P SULLIVAN W REINECKE I MCNAB 4030 2 W BRAKER LN AUSTIN TX 78759
1	INTEGRATED COMPOSITE TECH H PERKINSON JR PO BOX 397 YORK NEW SALEM PA 17371-0397
1	INTERFEROMETRICS INC R LARRIVA VICE PRESIDENT 8150 LEESBURG PIKE VIENNA VA 22100
1	AEROSPACE RES & DEV (ASRDD) CORP D ELDER PO BOX 49472 COLORADO SPRINGS CO 80949-9472
1	PM ADVANCED CONCEPTS LORAL VUGHT SYSTEMS J TAYLOR MS WT 21 PO BOX 650003 DALLAS TX 76265-0003

NO. OF
COPIES ORGANIZATION

2 LORAL VOUGHT SYSTEMS
G JACKSON
K COOK
1701 W MARSHALL DR
GRAND PRAIRIE TX 75051

1 BRIGS CO
J BACKOFEN
2668 PETERBOROUGH ST
HERDON VA 22071-2443

1 SOUTHWEST RSCH INSTITUTE
J RIEGEL
ENGR & MATL SCIENCES DIV
6220 CULEBRA RD
PO DRAWER 28510
SAN ANTONIO TX 78228-0510

1 ZERNOW TECHNICAL SERVICES
L ZERNOW
425 W BONITA AVE SUITE 208
SAN DIMAS CA 91773

1 R EICHELBERGER CONSULTANT
409 W CATHERINE ST
BEL AIR MD 21014-3613

1 DYNA EAST CORPORATION
P CHI CHOU
3201 ARCH ST
PHILADELPHIA PA 19104-2711

2 MARTIN MARIETTA CORP
P DEWAR
L SPONAR
230 EAST GODDARD BLVD
KING OF PRUSSIA PA 19406

2 OLIN CORPORATION
FLINCHBAUGH DIV
E STEINER
B STEWART
PO BOX 127
RED LION PA 17356

1 OLIN CORPORATION
L WHITMORE
10101 9TH ST NORTH
ST PETERSBURG FL 33702

NO. OF
COPIES ORGANIZATION

1 RENNSAELER POLYTECHNIC
INSTITUTE
R B PIPES
PRESIDENT OFC PITTSBURGH BLDG
TROY NY 12180-3590

1 SPARTA INC
J GLATZ
9455 TOWNE CTR DRIVE
SAN DIEGO CA 92121-1964

2 UNITED DEFENSE LP
P PARA
G THOMAS
1107 COLEMAN AVE BOX 367
SAN JOSE CA 95103

1 MARINE CORPS SYSTEMS COMMAND
PM GROUND WPNS
COL R OWEN
2083 BARNETT AVE SUITE 315
QUANTICO VA 22134-5000

1 OFFICE OF NAVAL RES
J KELLY
800 NORTH QUINCEY ST
ARLINGTON VA 22217-5000

2 NAVAL SURFACE WARFARE CTR
CARDEROCK DIVISION
R CRANE CODE 2802
C WILLIAMS CODE 6553
3A LEGGETT CIR
ANNAPOLIS MD 21402

5 SIKORSKY
H BUTTS
T CARSTENSAN
B KAY
S GARBO
J ADELMANN
6900 MAIN ST
PO BOX 9729
STRATFORD CT 06601-1381

1 U WYOMING
D ADAMS
PO BOX 3295
LARAMIE WY 82071

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
1	MICHIGAN ST UNIVERSITY R AVERILL 3515 EB MSM DEPT EAST LANSING MI 48824-1226
1	AMOCO POLYMERS J BANISAUKAS 4500 MCGINNIS FERRY RD ALPHARETTA GA 30005
1	HEXCEL T BITZER 11711 DUBLIN BLVD DUBLIN CA 94568
1	BOEING R BOHLMANN PO BOX 516 MC 5021322 ST LOUIS MO 63166-0516
1	NAVSEA OJRI G CAMPONESCHI 2351 JEFFERSON DAVIS HWY ARLINGTON VA 22242-5160
1	LOCKHEED MARTIN R FIELDS 1195 IRWIN CT WINTER SPRINGS FL 32708
1	USAF WL MLS OL A HAKIM 5225 BAILEY LOOP 243E MCCLELLAN AFB CA 55552
1	PRATT & WHITNEY D HAMBRICK 400 MAIN ST MS 114-37 EAST HARTFORD CT 06108
1	BOEING DOUGLAS PRODUCTS DIV L J HART-SMITH 3855 LAKEWOOD BLVD D800-0019 LONG BEACH CA 90846-0001

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
1	MIT P LAGACE 77 MASS AVE CAMBRIDGE MA 01887
1	NASA-LANGLEY J MASTERS MS 389 HAMPTON VA 23662-5225
1	CYTEC M LIN 1440 N KRAEMER BLVD ANAHEIM CA 92806
2	BOEING ROTORCRAFT P MINGURT P HANDEL 800 B PUTNAM BLVD WALLINGFORD PA 19086
2	FAA TECH CENTER D OPLINGER AAR-431 P SHYPRYKEVICH AAR-431 ATLANTIC CITY NJ 08405
1	NASA-LANGLEY RC C C POE MS 188E NEWPORT NEWS VA 23608
1	LOCKHEED MARTIN S REEVE 8650 COBB DR D 73 62 MZ 0648 MARIETTA GA 30063-0648
1	WL MLBC E SHINN 2941 PST STE 1 WRIGHT PAT AFB OH 45433-7750
2	IIT RESEARCH CENTER D ROSE 201 MILL ST ROME NY 13440-6916
1	MATERIALS SCIENCES CORP B W ROSEN 500 OFFICE CENTER DR STE 250 FORT WASHINGTON PA 19034

NO. OF
COPIES ORGANIZATION

1 DOW UT
S TIDRICK
15 STERLING DR
WALLINGFORD CT 06492

3 TUSKEGEE UNIVERISTY
MATERIALS RESEARCH LAB
SCHOOL OF ENGR & ARCH
S JEELANI
H MAHFUZ
U VAIDYA
TUSKEGEE AL 36088

4 NIST
POLYMERS DIVISION
R PARNAS
J DUNKERS
M VANLANDINGHAM
D HUNSTON
GAITHERSBURG MD 20899

2 NORTHROP GRUMMAN
ENVIRONMENTAL PROGRAMS
R OSTERMAN
8900 E WASHINGTON BLVD
PICO RIVERA CA 90660

1 OAK RIDGE NATL LAB
A WERESZCZAK
BLDG 4515 MS 6069
PO BOX 2008
OAKRIDGE TN 37831-6064

1 COMMANDER
USARDEC
INDUSTRIAL ECOLOGY CTR
T SACHAR
BLDG 172
PICATINNY ARSENAL NJ
07806-5000

1 COMMANDER
USA AMCOM
AVIATION APPLIED TECH DIR
J SCHUCK
FT EUSTIS VA

NO. OF
COPIES ORGANIZATION

1 COMMANDER
US ARMY ARDEC
AMSTA AR SRE D YEE
PICATINNY ARSENAL NJ
07806-5000

11 COMMANDER
US ARMY ARDEC
AMSTA AR CCH B
B KONRAD
E RIVERA
G EUSTICE
S PATEL
G WAGNECZ
R SAYER
F CHANG
M BOWMAN
P VALENTI
L MANOLE
W RICE
BLDG 65
PICATINNY ARSENAL NJ
07806-5000

1 COMMANDER
US ARMY ARDEC
AMSTA AR QAC T D RIGOGLIOSO
BLDG 354 M829E3 IPT
PICATINNY ARSENAL NJ
07806-5000

5 DIRECTOR
US ARMY RESEARCH LAB
AMSRL WM MB
A ABRAHAMIAN
M BERMAN
A FRYDMAN
T LI
W MCINTOSH
2800 POWDER MILL RD
ADELPHI MD 20783-1145

NO. OF
COPIES ORGANIZATION

ABERDEEN PROVING GROUND

70 DIR USARL
AMSRL CI
AMSRL CI HA
W STUREK
A MARK
AMSRL IS CD
R KASTE
AMSRL SL B
AMSRL SL BA
AMSRL SL BE
D BELY
AMSRL WM B
A HORST
E SCHMIDT
AMSRL WM BE
G WREN
C LEVERITT
D KOOKER
AMSRL WM BC
P PLOSTINS
D LYON
J NEWILL
K SOENCKSEN
S WILKERSON
AMSRL WM BD
R FIFER
B FORCH
R PESCE RODRIGUEZ
B RICE
AMSRL WM M
D VIECHNICKI
G HAGNAUER
J MCCAULEY
AMSRL WM MA
R SHUFORD
S MCKNIGHT
N BECK TAN
L GHIORSE
P MOY
AMSRL WM MB
B BURNS
W DRYSDALE
J BENDER
T BLANAS
T BOGETTI
R BOSSOLI
L BURTON
S CORNELISON

NO. OF
COPIES ORGANIZATION

AMSRL WM MB (CONTINUED)
P DEHMER
R DOOLEY
B FINK
G GAZONAS
S GHIORSE
D GRANVILLE
D HOPKINS
C HOPPEL
D HENRY
R KASTE
M LEADORE
R LIEB
E RIGAS
D SPAGNUOLO
W SPURGEON
E SZYMANSKI
J TZENG
AMSRL WM MC
J BEATTY
AMSRL WM MD
W ROY
AMSRL WM TA
W GILLICH
E RAPACKI
T HAVEL
AMSRL WM TC
R COATES
W DE ROSSET
AMSRL WM TD
D DIETRICH
W BRUCHEY
A DAS GUPTA
AMSRL WM BB
H ROGERS
AMSRL WM BA
F BRANDON
W D AMICO
AMSRL WM BR
J BORNSTEIN
AMSRL WM TE
A NIILER
AMSRL WM BF
J LACETERA

REPORT DOCUMENTATION PAGE			Form Approved OMB No. 0704-0188	
Public reporting burden for this collection of information is estimated to average 1 hour per response, including the time for reviewing instructions, searching existing data sources, gathering and maintaining the data needed, and completing and reviewing the collection of information. Send comments regarding this burden estimate or any other aspect of this collection of information, including suggestions for reducing this burden, to Washington Headquarters Services, Directorate for Information Operations and Reports, 1215 Jefferson Davis Highway, Suite 1204, Arlington, VA 22202-4302, and to the Office of Management and Budget, Paperwork Reduction Project(0704-0188), Washington, DC 20503.				
1. AGENCY USE ONLY (Leave blank)	2. REPORT DATE September 1999	3. REPORT TYPE AND DATES COVERED Final, January 1997-May 1999		
4. TITLE AND SUBTITLE Evaluation of Obturator and Sealing Cuff Properties for the M865 Training Projectile With Comparison to Ballistic Testing		5. FUNDING NUMBERS 622618AH80		
6. AUTHOR(S) C. P. R. Hoppel, J. F. Newill, and K. P. Soencksen				
7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES) U.S. Army Research Laboratory ATTN: AMSRL-WM-MB Aberdeen Proving Ground, MD 21005-5069		8. PERFORMING ORGANIZATION REPORT NUMBER ARL-TR-2039		
9. SPONSORING/MONITORING AGENCY NAMES(S) AND ADDRESS(ES)		10. SPONSORING/MONITORING AGENCY REPORT NUMBER		
11. SUPPLEMENTARY NOTES				
12a. DISTRIBUTION/AVAILABILITY STATEMENT Approved for public release; distribution is unlimited.		12b. DISTRIBUTION CODE		
13. ABSTRACT (Maximum 200 words) The nylon obturator and RTV sealing cuff for the M865 training round were evaluated to identify potential sources of ballistic variability associated with the material properties and material processing. While the properties of these materials are strongly dependent on processing conditions, temperature, and moisture content, the M865 performance variability is reduced by a well-engineered fracture mechanism that focuses the stresses in the obturator during sabot discard. A ballistic test was developed to validate the study. For the ballistic test, obturators were manufactured in "brittle," "tough," and "tough-wet" conditions. These three conditions produced significant differences in the mechanical properties (the maximum strength varied by a factor of 2, the elastic modulus varied by a factor of 25, and the elongation to failure varied by a factor of 10). However, the ballistic performance did not show any significant variability due to the obturator properties.				
14. SUBJECT TERMS obturator, nylon, kinetic energy, M865, training, projectile		15. NUMBER OF PAGES 41		
		16. PRICE CODE		
17. SECURITY CLASSIFICATION OF REPORT UNCLASSIFIED	18. SECURITY CLASSIFICATION OF THIS PAGE UNCLASSIFIED	19. SECURITY CLASSIFICATION OF ABSTRACT UNCLASSIFIED	20. LIMITATION OF ABSTRACT UL	

INTENTIONALLY LEFT BLANK.

USER EVALUATION SHEET/CHANGE OF ADDRESS

This Laboratory undertakes a continuing effort to improve the quality of the reports it publishes. Your comments/answers to the items/questions below will aid us in our efforts.

1. ARL Report Number/Author ARL-TR-2039 (Hoppel) Date of Report September 1999
2. Date Report Received _____
3. Does this report satisfy a need? (Comment on purpose, related project, or other area of interest for which the report will be used.) _____

4. Specifically, how is the report being used? (Information source, design data, procedure, source of ideas, etc.) _____

5. Has the information in this report led to any quantitative savings as far as man-hours or dollars saved, operating costs avoided, or efficiencies achieved, etc? If so, please elaborate. _____

6. General Comments. What do you think should be changed to improve future reports? (Indicate changes to organization, technical content, format, etc.) _____

CURRENT
ADDRESS

Organization

Name

E-mail Name

Street or P.O. Box No.

City, State, Zip Code

7. If indicating a Change of Address or Address Correction, please provide the Current or Correct address above and the Old or Incorrect address below.

OLD
ADDRESS

Organization

Name

Street or P.O. Box No.

City, State, Zip Code

(Remove this sheet, fold as indicated, tape closed, and mail.)
(DO NOT STAPLE)